

# Section 1

IMPACT TESTING SETUP AND RIGID IMPACTOR STUDY

Dan Duke, Ph.D. P.E.

TRIDYNAMIC SOLUTIONS

# Recommendations

- ▶ Require impact using soft impactors.
- ▶ Soft impactors shall have characteristics similar to the wing of a small aircraft.

# Rigid vs Soft Impactors

- ▶ Rigid Impactors tend to yield energy measurements that are lower than soft impactors.
- ▶ Energy measurements are not equivalent over the contact period.

# Rigid vs Soft Impactors

## Simulation Results

5

| Device Type       | Maximum Energy (kJ-m) |      | Comment                                                          |
|-------------------|-----------------------|------|------------------------------------------------------------------|
|                   | Rigid                 | TC2  |                                                                  |
| Aluminum Lattice  | 13.6                  | 16.7 | Variation in Energy is the Result of Variations in Failure Modes |
| Aluminum Pipe     | 23.6                  | 17.2 |                                                                  |
| Composite Lattice | 14.8                  | 18.5 |                                                                  |
| Composite Pipe    | 4.6                   | 14.2 |                                                                  |

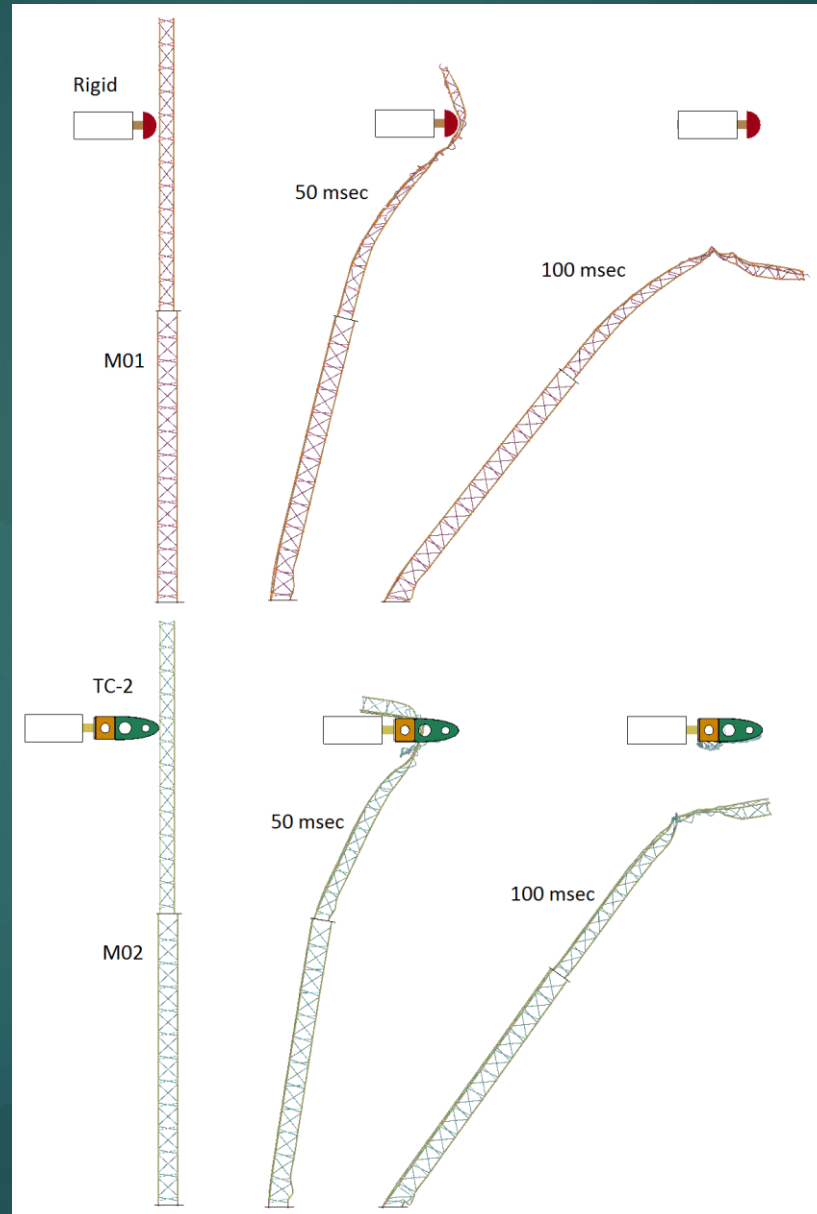
Load Cell Energy  
Lower in 3 of 4 Cases

68% Low

# Rigid vs Soft Impactors

## *Tower Response*

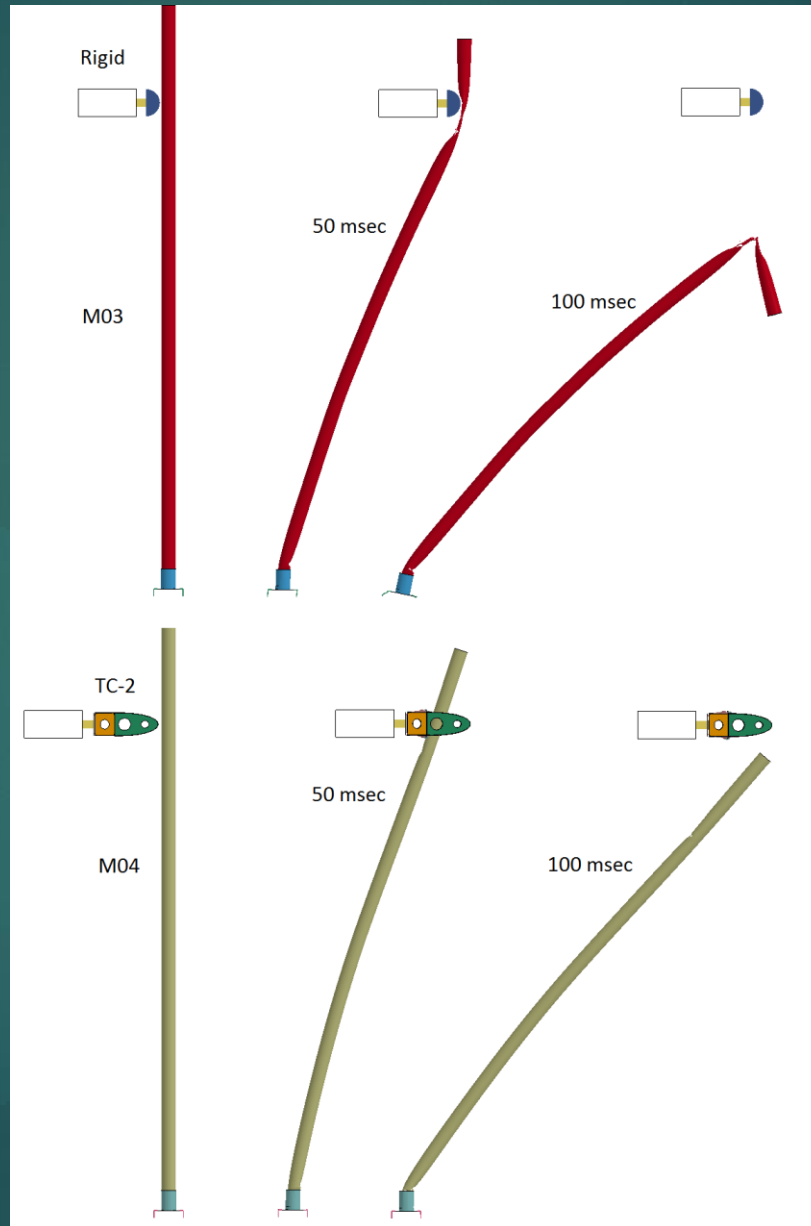
6



# Rigid vs Soft Impactors

## *Tower Response*

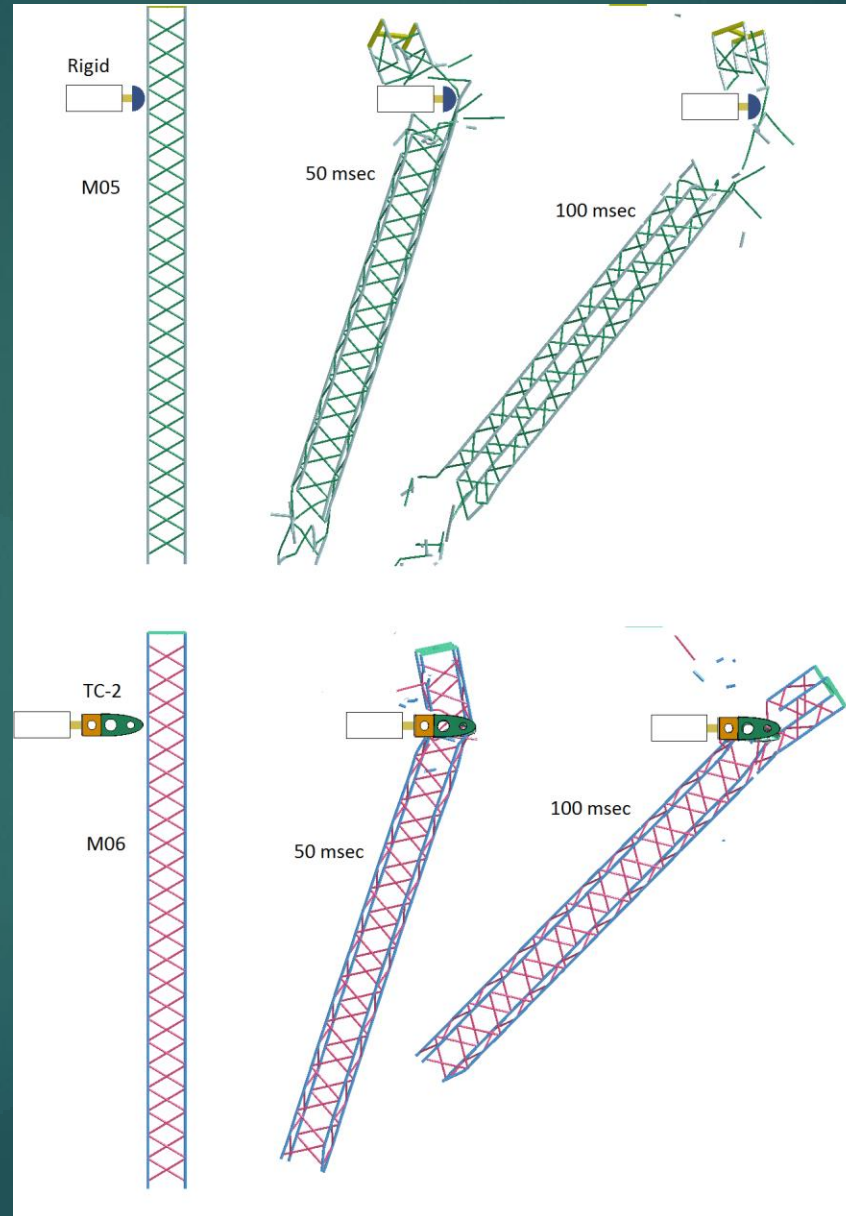
7



# Rigid vs Soft Impactors

## *Tower Response*

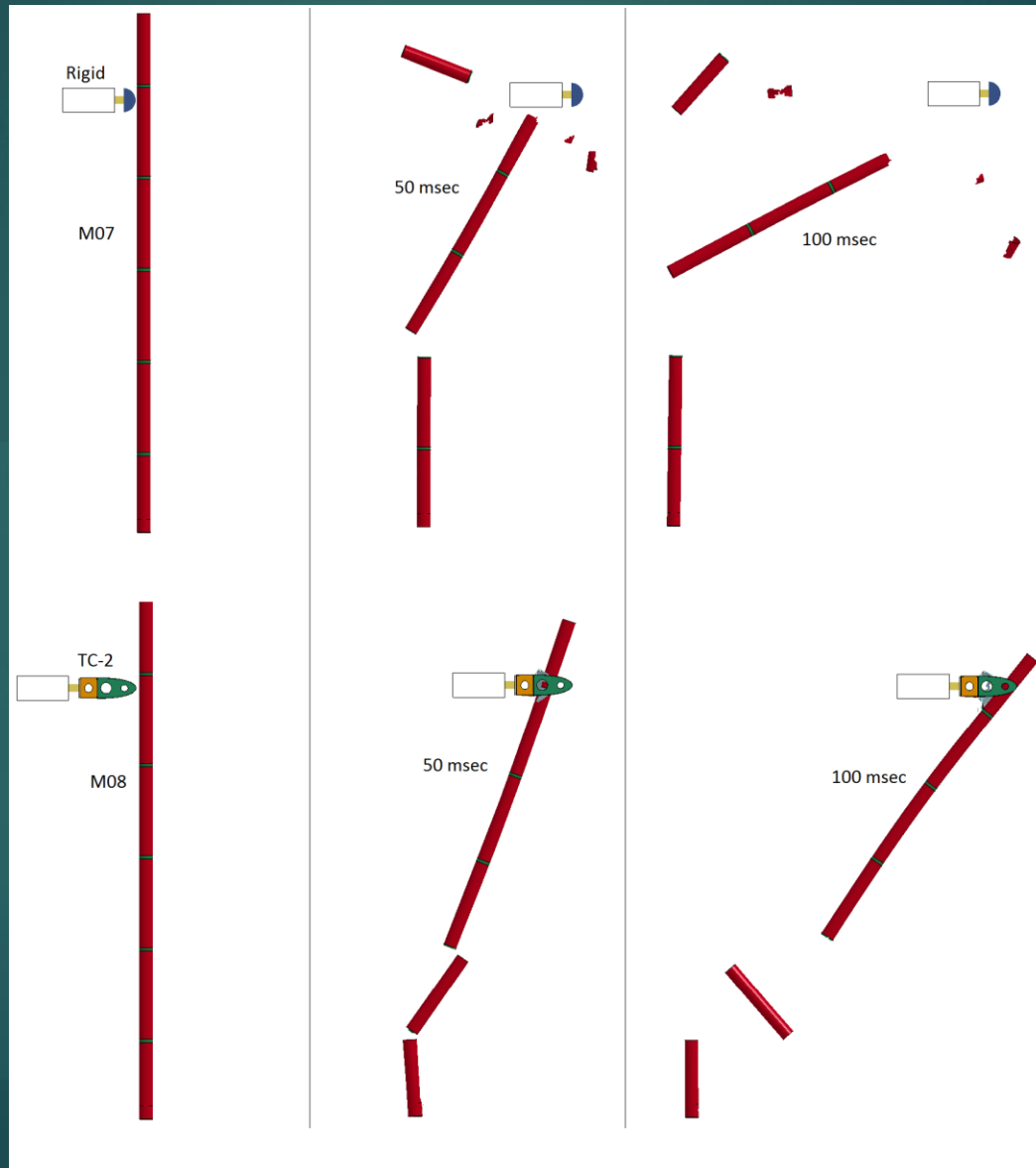
8



# Rigid vs Soft Impactors

## *Tower Response*

9



# Rigid vs Soft Impactors

## *Historical Test Results*

10

- ▶ Contact Period and Energy
  - ▶ Significant Difference

| Measurement                       | Rigid<br>(avg of 3) | TC2<br>(avg of 2) | %<br>Difference |
|-----------------------------------|---------------------|-------------------|-----------------|
| Contact period (msec)             | 59.3                | 85                | <b>43%</b>      |
| Energy over contact period (kN-m) | 13.0                | 15.9              | <b>22%</b>      |

# Rigid vs Soft Impactors

## *Historical Test Results*

11

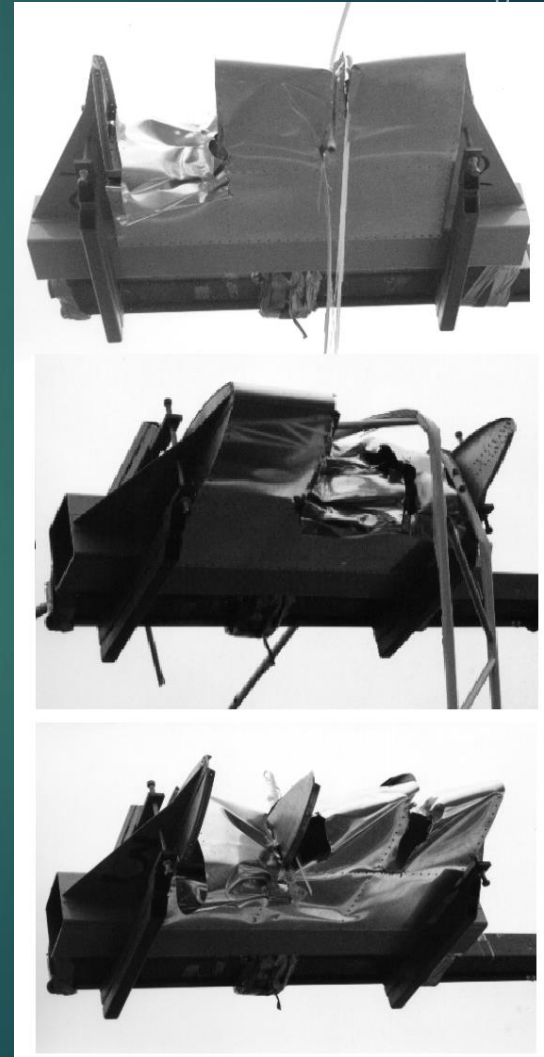
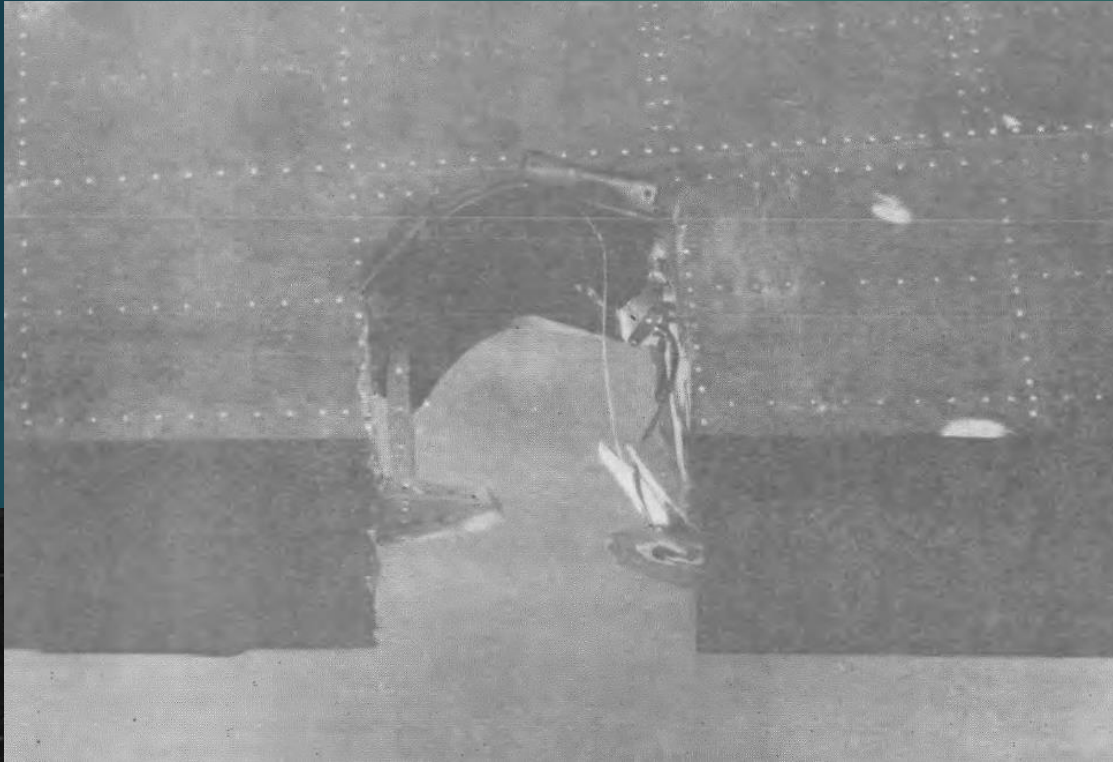
### ► Different Failure Mechanisms

| Measurement                      | Rigid<br>(avg of 3) | TC2<br>(avg of 2) | %<br>Difference |
|----------------------------------|---------------------|-------------------|-----------------|
| Time to failure: First (msec)    | 14.7                | 32.0              | <b>118%</b>     |
| Time to failure: Second (msec)   | 32.0                | n/a               | n/a             |
| Energy to failure: First (kN-m)  | 5.01                | 10.7              | <b>114%</b>     |
| Energy to failure: Second (kN-m) | 10.1                | n/a               | n/a             |

# Rigid Impactors

*Lost Value of Visual Inspection*

12



# Rigid vs Soft Impactors

## *Summary*

13

- ▶ Rigid impactors do not yield higher energy values than soft wing surrogates.
- ▶ Tower responses and failure modes are very different for rigid versus soft impactors.
- ▶ Rigid impactors do not support visual inspection of wing damage.
- ▶ Impactors similar to the wings of a small aircraft are simply more realistic.

# Shane Shurtliff, P.E.

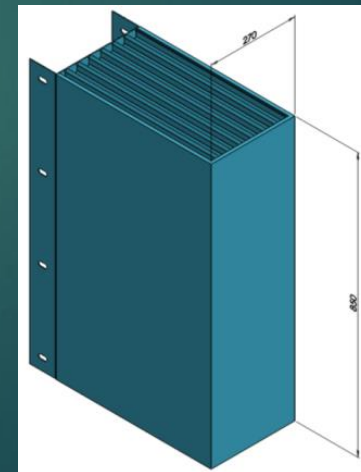
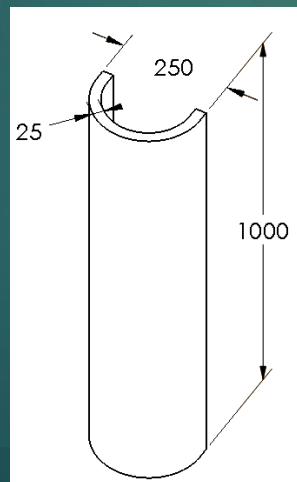
SELECT ENGINEERING SERVICES

# Rigid vs Soft Impactors

## *Impactor Design*

15

- ▶ Using a rigid impactor does not produce the same results as a deformable impactor such as an airplane wing.
- ▶ To use a deformable impactor, it must be repeatable in order to establish a standard.
- ▶ Crush Strength would be designed to represent an aircraft wing.
- ▶ Honeycomb impactor are repeatable, customizable, and inexpensive to produce.
- ▶ Recommend using honeycomb impactor

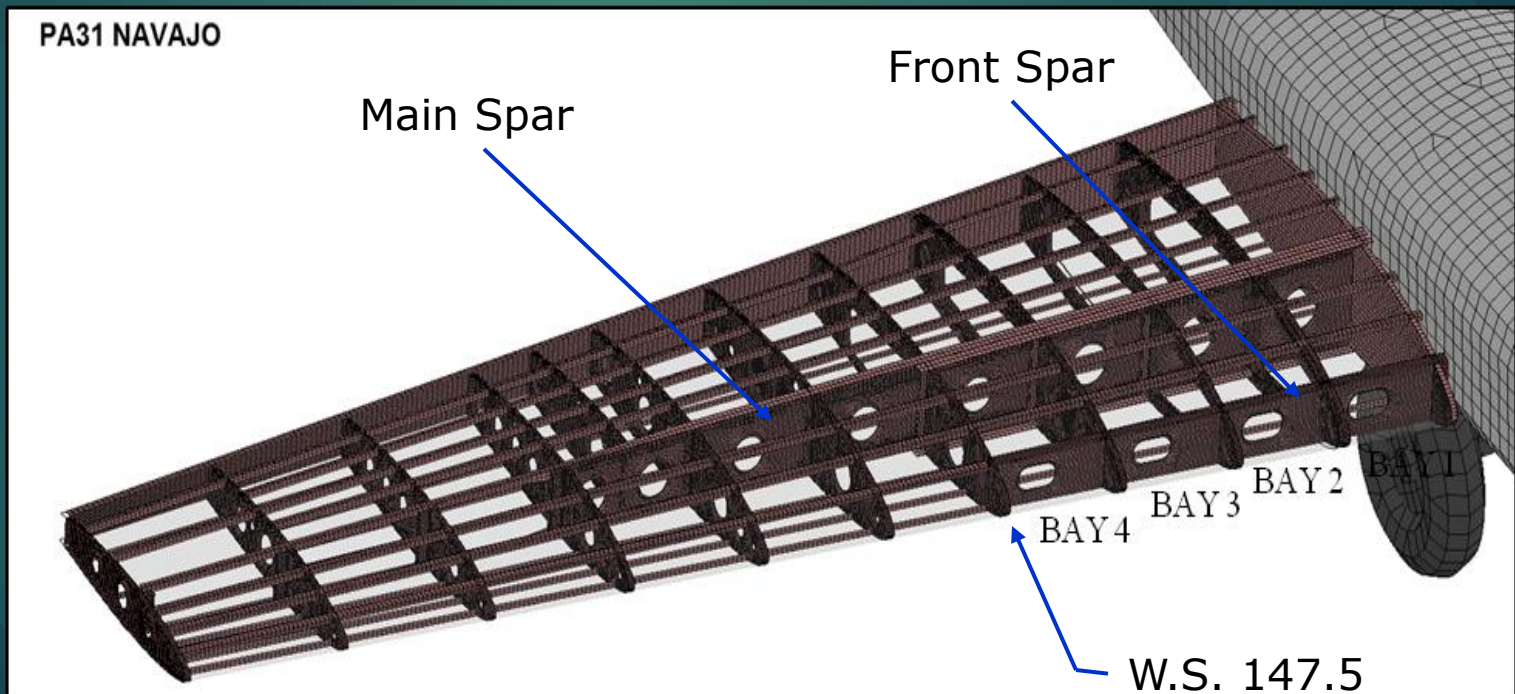


# Rigid vs Soft Impactors

## *Impactor Design*

16

- ▶ Used previously for FAA Tests
- ▶ Close to 3000kg weight
- ▶ Able to obtain full drawing package to generate computer model



# Rigid vs Soft Impactors

## *Impactor Design*

17

- ▶ Performed static crush tests on three different designs
- ▶ Determined which design to use based on crush forces as compared to wing data
- ▶ Performed dynamic impacts using drop tower system
- ▶ Standard design allows test data between different products to be compared

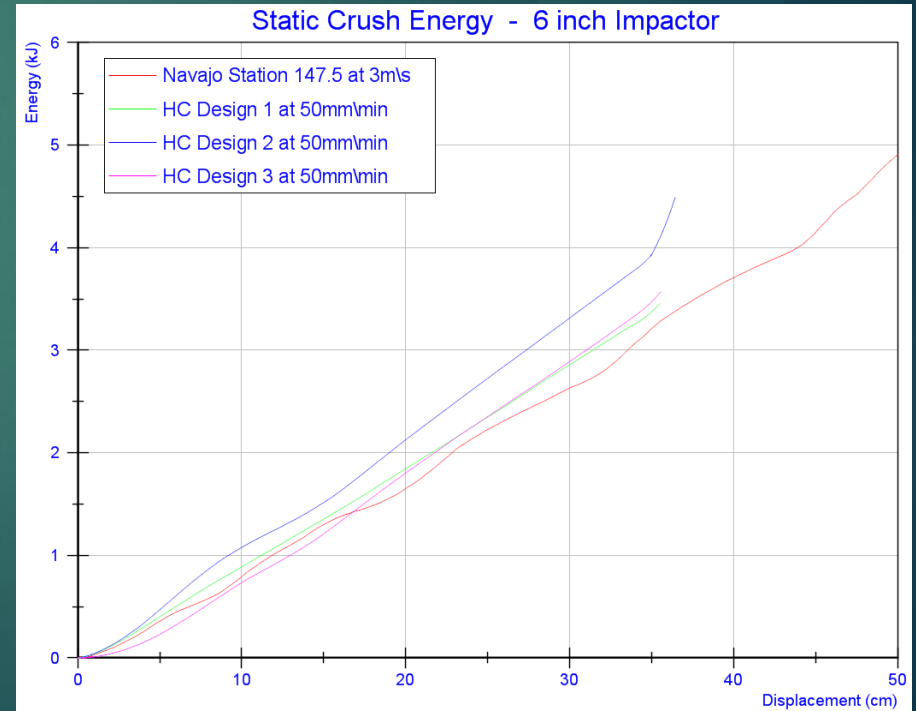
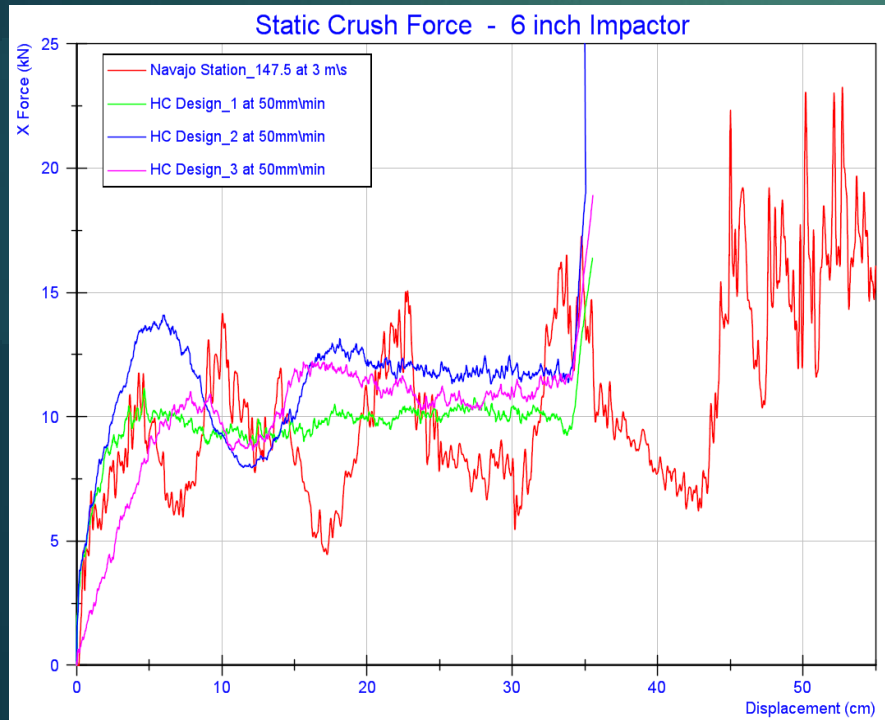


# Rigid vs Soft Impactors

## Impactor Design

18

- ▶ Crush Strength of Honeycomb compared to crush strength of wing
- ▶ Good match to Piper Navajo



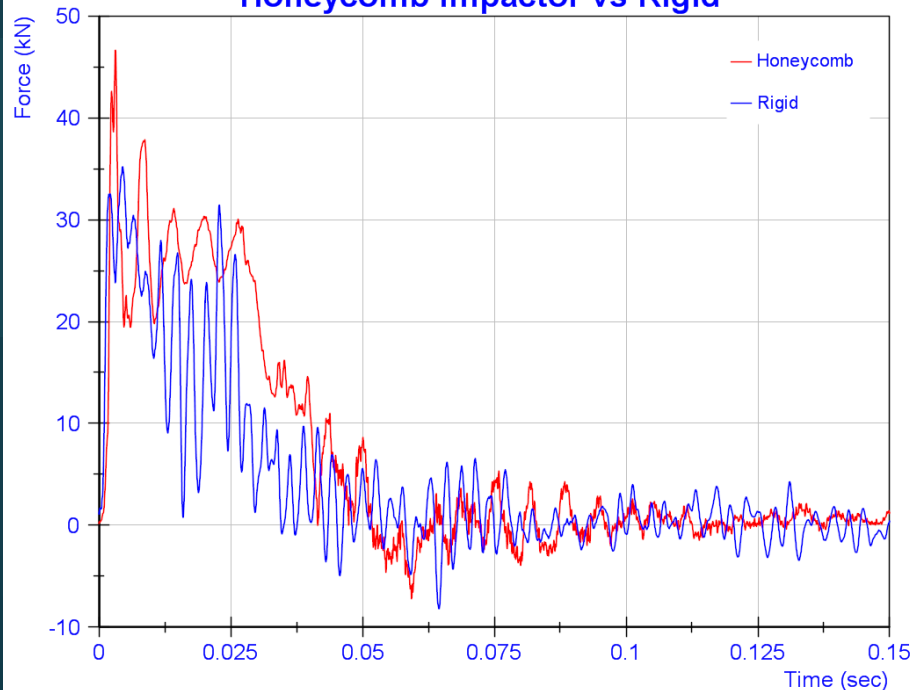
# Rigid vs Soft Impactors

## *Impactor Design*

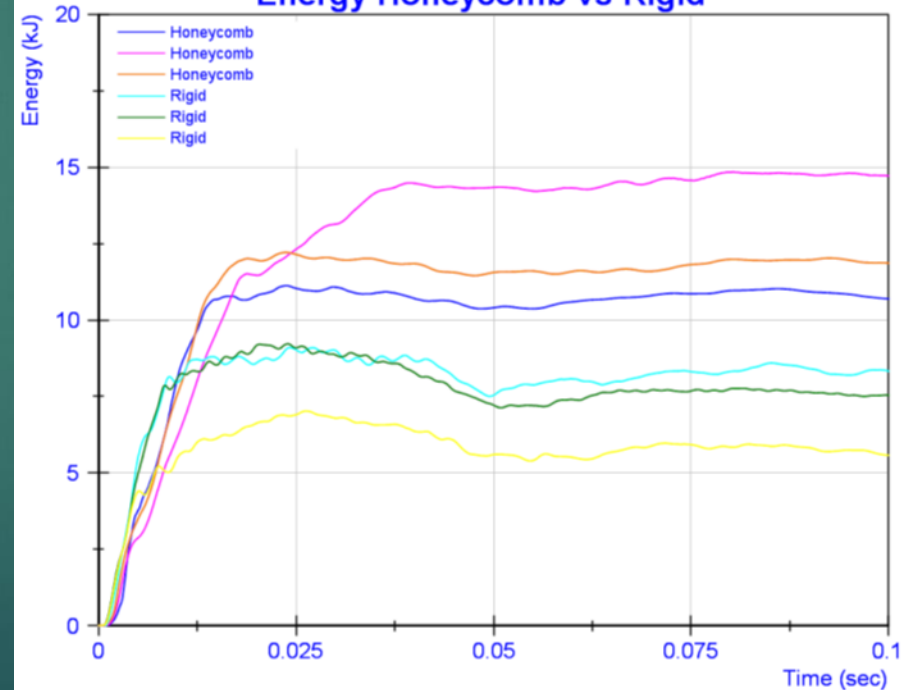
19

- ▶ The rigid impactor generated more noise in the data
- ▶ Significantly reduced the energy required to break through the pole.
- ▶ Changes failure mode

**Honeycomb Impactor vs Rigid**



**Energy Honeycomb vs Rigid**



# Rigid vs Soft Impactors

## *Impactor Design*

20

### ► Soft vs Rigid Impactor

- The rigid impactor does not provide energy values representative of an airplane wing.

| Calculated Energy Values for FAA Impact Tests |                             |               |              |
|-----------------------------------------------|-----------------------------|---------------|--------------|
|                                               | Rigid Impactor              | Soft Impactor | % Difference |
| Product X                                     | 8.43                        | 12.60         | 33.07        |
| Product Y                                     | 25.67                       | 36.43         | 29.55        |
| Product Z                                     | 25.50                       | 42.67         | 40.23        |
|                                               | <b>Average % Difference</b> |               | <b>34.28</b> |

# Rigid vs Soft Impactors

## *Impactor Design*

21

- ▶ Rigid impactor causes more localized failure in the LIR structure

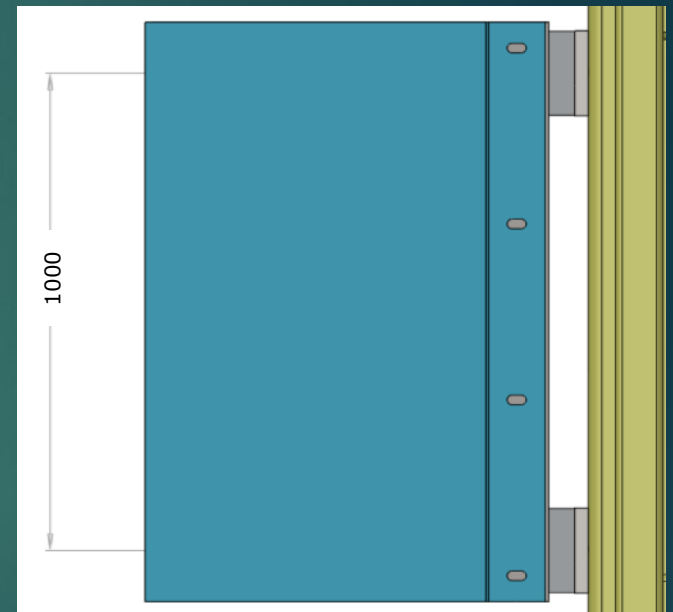


# Summary and Recommendations

## *Impactor Design and Instrumentation*

22

- ▶ Standard Honeycomb Impactor
- ▶ Tri-axial load cells
- ▶ Max two load cells per impactor
- ▶ Load cell spacing should be no larger than 1 meter
- ▶ Minimize weight in front of load cells (no greater than 55 pounds (25 kg))
- ▶ Record data at a min of 10 kHz
- ▶ Use High Speed video at a minimum of 1000 fps
- ▶ Video must capture failure mode and duration of impact

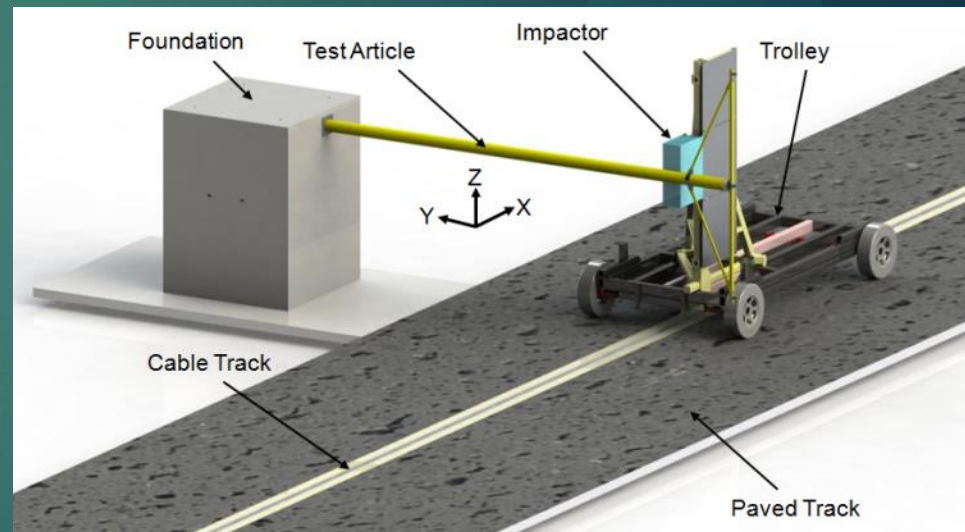


# Summary and Recommendations

## *Test System and Setup*

23

- ▶ Define stiffness requirement for structure behind impactor.
- ▶ Test article may be mounted horizontal or vertical.
- ▶ The X-axis is defined as the direction of impact.
- ▶ Impact location 1 meter from top (research needed for other impact locations)
- ▶ Standardize Pole Length



# Recommendations

24

- ▶ Require impact using soft impactors.
- ▶ Soft impactors shall have characteristics similar to the wing of a small aircraft.